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MEMORANDUM FOR RECORD

16 September 1954

SUBJECT: Meeting of Upper Atmosphere Rocket Research Panel

1. At the invitation of Air Force Scientific Advisory Board, I attended a meeting of the Upper Atmosphere Rocket Research Panel on 8 September at the Naval Research Laboratory, Anacostia. The two agenda items of primary concern were: 1) Higher Altitude and Satellite Vehicles, 2) International Geophysical Year, and the logistics therefor.

2. This panel was formed in 1943 by representatives of various organizations which were concerned with upper atmosphere research through rocketry. It is chaired by Dr. James A. Van Allen who is Chairman of the Department of Physics, State University of Iowa. The panel has representation from Aberdeen Proving Grounds, University of Michigan, Naval Signal Laboratory, General Electric Company, Naval Research Laboratory, Air Force Cambridge Research Center, California Institute of Technology, and Harvard Observatory. It has no official connection with any department of defense or education but is informally supported by the Office of Naval Research. In connection with the forthcoming International Geophysical Year, it has been assigned the function of developing the United States program for upper atmosphere research through the use of high altitude rockets.

3. At the 8 September meeting, there were in addition to the panel itself, representatives from the Chief of Ordnance, United States Army, the Ordnance Arsenal, The Rand Corporation, the Office of Naval Research, Aerophysics Development Corporation, Air Force AFOSR, AFOSR, National Science Foundation, etc.

4. Dr. Van Allen opened the discussion of the first agenda item - Higher Altitude and Satellite Vehicles - by a discussion of past upper atmosphere research with rockets. He cited the extensive use of V-2s which could carry 2,500 pounds payload to a 100 miles altitude but the supply of which has now been exhausted. He mentioned the WAC Corporal which was not used as a single stage vehicle, primarily because it could carry only 25 pounds to an altitude of 80 miles. The WAC Corporal was however, used as a second stage in combination with the V-2 and achieved an altitude of 250 miles. Since exhaustion of the V-2 supply, the other rockets have been developed and used as civilian research instruments. These are the Viking which in its present form can carry 100 pounds payload to 135 miles

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4. Dr. Van Allen opened the discussion of the first agenda item - Higher Altitude and Satellite Vehicles - by a discussion of past upper atmosphere research with rockets. He cited the extensive use of V-2s which could carry 2,500 pounds payload to a 150 mile altitude but the supply of which has now been exhausted. He mentioned the WAC Corporal which was not used as a single stage vehicle, primarily because it could carry only 2 1/2 pounds to an altitude of 50 miles. The WAC Corporal was however, used as the second stage in combination with the V-2 and achieved an altitude of 250 miles. Since exhaustion of the V-2 supply, the other rockets have been developed and used as civilian research instruments. These are the Viking which in its present version can carry 100 pounds payload to 135 miles

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altitude and in its forthcoming modification will carry 500 pounds to 195 miles altitude. The second rocket is the Aerobee which can carry 150 pounds payload to 65 miles altitude. In addition to these two, there is also the Black Arrow which is a system of launching a rocket from a "gondola" balloon. The launch is made at a balloon altitude of approximately 100,000 feet and the rocket carries 30 pounds of instrumentation to an altitude of 60 miles. Mr. Van Allen concluded his presentation by stating that, from here on, civilian upper atmosphere rocket research will probably be dependent, due to lack of civilian funds, upon the military rockets being developed now by the Department of Defense. He then introduced Mr. Fred Hinkle of the Harvard Observatory for a discussion of North Satellite Vehicle (NSV).

5. Mr. Hinkle stated that the interests of civilian research and of the Department of Defense in NSV are synonymous. He pointed out the main scientific aims which could be made through the NSV which have application both to civilian science and to defense. These include meteorology, astronomy, solar-terrestrial relations, far ultra-violet and X-ray research, etc. He described the NSV sequence as being in three phases as follows:

(1) The placing in orbit of an observable object (uninstrumented) which could be seen either optically or by radar.

(2) The placing in orbit of an instrumented vehicle, "an unmanned physical laboratory". This phase will be a progressive development starting with a very small vehicle similar to the Singer, which will carry telemetering equipment. Larger capacity will be achieved as bigger and more reliable power plants are developed. These latter will probably be nuclear or solar energy plants. Instrumentation starting with simple telemetering will progress through more complicated stages until television and finally a telescope is included. The last stage in this phase will be a remote controlled vehicle.

(3) The final phase of NSV will be the putting in orbit of a manned satellite vehicle.

6. Mr. Hinkle emphasized that the placing of a slug in orbit was in Phase One first step in the process. He stated that such a vehicle even without instrumentation could produce useful scientific results such as air density data and relative positions on the earth. The main problem in connection with a slug (Phase II) is that of observation from the earth, which will require much study. This problem will be simplified if the NSV is on either an equatorial or a polar orbit. In regard to

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the unmanned instrumented vehicle (Phase 2), he said that the main problems will be development of small reliable power plants; television, the technical design of which is already well along; orientation of the vehicle itself and of the instrumentation carried; and constant reduction of weight of the equipment to be carried. He stated that the altitude at which the vehicle should orbit will depend upon the purpose envisaged. Theoretically, an altitude of 1,000 miles at a speed of 5 miles per second would be ideal. This would provide a 2-hour orbit.

7. Mr. George Hoover of the Air Branch, ONI, made the next presentation, concerning high altitude vehicle projects with which ONI is concerned. ONI has three main projects in this field, the first of which is the development of a manned conventional aircraft to operate at a maximum altitude of 300 miles. Two designs are presently under consideration - the Douglas 553 and the Douglas 555. The former is designed to have an altitude of 700,000 feet. (Note: this field of study is covered in report N-OT-1260(30), "1 in Altitude and High Speed Study" by Douglas Aircraft Corporation - ONI-127141). The second ONI project is development of a manned high altitude balloon to operate at 100 to 200,000 feet. The system would be based on the "Skyhook" polyethylene balloon carrying a gondola equipped to sustain one or two men. The third ONI project is for manned space flight and the study of this is being conducted at the Aero Jet Laboratory. The first phase of this latter project is called S103, the purpose of which is to place an RV in orbit at an altitude of 200 miles in order to secure meteorological and atmospheric data at that level as the first step toward higher altitude work. Project S103 has been tentatively approved by the Navy and ONI is going ahead with it in cooperation with the Army. It is now also being coordinated with the Air Force at a very high level. The project calls for the use of the Army Redstone missile (see para. 8 below) as the first stage with the Loki cluster (see para. 9 below) providing the second and third stages. Under Aero Jet leadership, four subsidiary studies are planned to be undertaken as follows:

- (1) A Feasibility Study to determine the size and weight of the vehicle required at an altitude of 300 miles. It is hoped that Mr. Fred Whipple will be the leader of this.
- (2) An Orbital Study to determine the power required, the guidance system, etc. It is hoped that Mr. S. Fred Singer of the University of Maryland will lead this.
- (3) A Launch-trajectory Study to determine the final design and the staging requirements.

(h) A launching study to determine where and how the vehicle should be launched, the logistics requirements, and the range risks involved.

Following the completion of these four studies, construction of the actual vehicle will be commenced. WAC considers that successful completion of Project WAC will lead into the launching of a vehicle similar to the WAC, i.e., an instrumented vehicle using a polar orbit at an altitude of 200 miles. It is expected that WAC would remain aloft for ten days while WAC could probably maintain its orbit for about one month. Mr. WAC never emphasized that if adequate backing is available for Project WAC, WAC might well be used during the International Geophysical Year. He laid great stress on the necessity for the United States being the first to launch an WAC and said that Project WAC was absolutely essential to achieve this end.

8. Mr. Miller of Redstone Arsenal then gave a briefing on the Redstone missile. The Redstone missile, as a tactical weapon, has a range of 150 miles with a trajectory apex of 45 miles. If used as a research tool, it has a vertical range of 150 miles with a weight of about 10,000 pounds. The modified WAC which is now under development will have a vertical range of 270 miles with an amount of weight allowance of 500 pounds. Redstone Arsenal is now working on the construction of these missiles as research instruments. The cost is 70,000 to 80,000 per missile if more than five or six are produced. They have a speed of approximately Mach 5.5.

9. Mr. Miller of Aerobics Development Corporation described the Aerobics test vehicle which they are developing as an air-borne rocket. The vehicle carries two clusters of four rockets; the first cluster of four acting as the first stage booster and the second cluster of four constituting the second stage. It has a speed of Mach 10 and a weight of 10,000 pounds. Air-launched vertically from an altitude of 40,000 feet, it is designed to achieve an altitude of 150 miles. It is planned to use either a flashing light system or a folding radar reflector for tracking purposes. When air-launched, it is expected to get an altitude of 150 miles. When test launched from the second stage develop 150 mts. The missile has yet to be flown vertically as a research instrument. Mr. Miller indicated that the second stage cluster could be adapted to carry 12 pounds of ruggedized telemetry equipment.

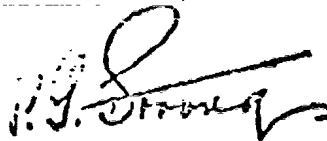
10. Mr. Miller of WAC presented briefly on a design which they are developing which consists of the WAC booster as the first stage and a second rocket as the second stage. It is estimated that this vehicle could attain an altitude of 100,000 feet. It would be very low cost, in the range of \$7,000 to \$13,000.

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11. Mr. William of the National Science Foundation gave a brief account of the situation of the funds for upper atmosphere rocket research. One million, three-hundred thousand dollars is presently available and about 70% of this will be transferred immediately to ONR for procurement and accounting purposes. The balance of the rocket project funds - \$400,000 - will become available next year.

12. Following adjournment of the panel meeting, I spent a couple of hours with Mr. Van Allen, who had served under me for a time during World War II. Van Allen was one of the key figures in the development of the W-2 space under Admiral Parsons and was one of the officers assigned to introduce the race to the Pacific Fleet. In our conversation, I mentioned the difficulty that apparently would be encountered in the JFV program, of designing scientific equipment which would stand the high G's encountered in most high altitude rockets, particularly with solid propellant. I mentioned specifically the 400 G's encountered in stages two of the Loki cluster vehicle. Mr. Van Allen said that there should be no such difficulty. He pointed out that the W-2 race contained five miniature radio tubes, a battery and a transmitter, and that this instrument, with essentially no failures, withstood 20,000 G's when fired from the rocket. He pointed out also that in the Loki cluster vehicle which has been in use for some time, the rocket motor carries 30 pounds of instrumentation for cosmic ray research and other scientific experiments at 40 G's. He said that he saw no difficulty whatsoever in designing instrumentation for high altitude rockets which would easily withstand 1,000 G's. In regard to the JFV program generally, he agreed thoroughly with Mr. McCreary, that the immediate first step was to launch a ship. He felt that if technical government priority could be established, with advance and control by an outside high level scientific group, it would almost certainly be possible to put a ship in orbit by the time of the JFV and possibly even put up an instrumented vehicle.

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